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Numerical Analysis of Cold-Formed Built-Up Battered Steel Columns Under Eccentric Compressive Loading

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Abstract- This paper presents a 3D finite element model simulation using ABAQUS finite element software to investigate the behavior of steel battered beam columns under uniaxial loads. The columns consist of double-channel cold-formed slender open sections. The study highlighted the effects of various factors on the ultimate capacity of these beam-column members.

By considering both global and local imperfections, the model focuses on the structural strength and performance of battered cold-formed steel columns made from built-up sections. These cold-formed steel (CFS) columns have a slenderness ratio ranging from 50 to 162. Axial loads are applied at the pin-ended support of the column, which comprises two cold-formed steel open-channel sections positioned at varying center-to-center distances.

To validate the finite element models, their results were compared with relevant experimental data from the literature on similar geometries. The finite element analysis accounted for various factors, including different geometrical properties of column sections, slenderness ratios, nonlinear material properties of cold-formed steel sections, built-up section components, and initial geometric imperfections.

Finally, the analysis results were compared to the predicted ultimate strengths according to both the North American Specification and European codes EC3. The findings show that the strength of battered columns is strongly influenced by the interaction of several factors, including the back-to-back distance of the two channels, depth-to-width ratios, number of battered, and overall column slenderness ratios.

Keywords- Finite Element Modeling, battered columns, back-to-back channels, Built-up column, Flexural buckling, Cold-formed steel, uniaxial loading, Direct Strength Method

I. INTRODUCTION

Recently, there has been an increase in the use of cold-formed steel (CFS) sections for various structural applications, largely due to their advantageous weight-to-strength ratios. This characteristic opens up opportunities for innovation and improved efficiency in construction processes. It also simplifies installation and fabrication while enabling the selection of cost-effective and efficient cross-sections [1-4].

Moreover, CFS sections are convenient and economical to handle and transport [5–6].

To enhance the structural performance of CFS built-up sections, chord elements are connected with lacings or battens along their lengths. These built-up sections are commonly used as columns to support significant axial and uniaxial loads. However, depending on the loading conditions, CFS members may experience different types of instability, including distortional buckling, flexural or overall (global) buckling, lateral-torsional buckling, and local buckling. Geometric imperfections can significantly affect the performance and ultimate capacity of CFS sections, as well as the effectiveness of strengthening measures [7-9].

Ongoing advancements in the development of design codes and recommendations for CFS structures have been driven by extensive research into improving the performance of CFS sections. Numerical simulations and experimental studies have been conducted on columns made from channel sections arranged in back-to-back and face-to-face configurations, often joined with batten plates connected by bolted, welded, or self-drilling screws [10-13].

Local web buckling in CFS sections with lower sectional compactness can be addressed using dividers between channel sections in built-up columns [14]. A detailed investigation is needed to determine the optimal spacer spacing for these types of columns. Laced built-up columns tend to offer superior axial strength and stability compared to battered columns, thanks to their continuous lateral connectivity [15]. Previous experimental studies on the cold-formed axial strength of built-up columns have been limited by a narrow range of geometric parameters [16-19]. The design strengths suggested by the Eurocode (EC3) [20] and North American specifications (NAS) [21] for short CFS built-up columns with lateral ties have often been considered overly conservative [22-25].

The novelty of this work lies in addressing the gap in research regarding the behaviour of channel built-up sections (CFS) under compression loads. While there has been extensive study on these members, there remains a limited investigation into their performance under varying levels of sectional

compactness, section slenderness, and unbraced chord slenderness. Therefore, this study is crucial in exploring the effects of back-to-back spacing and slenderness ratio on the axial compression and eccentric load performance of built-up columns, considering different sectional dimensions. Additionally, the work aims to evaluate the adequacy of current design standards in accurately predicting the uniaxial strengths of CFS battened built-up columns.

II. Finite Element Model

A. Modeling techniques

ABAQUS [26] was used to develop a numerical model simulating the behavior of a CFS battened column, accounting for both its elastic and plastic behavior. The following analysis section outlines the model's geometry and material properties. The goal of this model is to provide a reliable simulation of the performance of battened columns made from built-up cold-formed steel (BCFS). The self-drilling screws were modeled with the CONN3-D2 element, a three-dimensional beam connector with two nodes and six degrees of freedom per node [29, 30], as illustrated in Figure 1.

Buckling analysis using finite element methods is conducted in two stages. The first stage involves determining the buckling modes of the columns through Eigenvalue analysis. This linear elastic analysis uses the (*BUCKLE) command from the ABAQUS library, with the load applied in a specified step. Multiple buckling modes are evaluated, and the most relevant mode, as predicted by the Eigenvalue analysis, is selected. In the parametric study, the reference points are located at the column ends.

The second stage is a nonlinear load-displacement analysis that incorporates initial imperfections and material nonlinearity. This stage provides a detailed assessment of ultimate loads, failure modes, lateral displacements, axial strains, and axial shortenings.

B. Batten plates and chords of the channel

The S4R shell element is used to model the channels (chords) of slender cold-formed steel sections and batten plates. This element provides an accurate solution for most applications, with six degrees of freedom per node. The mesh was selected to ensure sufficient accuracy while minimizing computational time for modeling steel sections in cold-formed battened columns. To achieve this, an approximate global area of 25 mm² was chosen for elements sized 5 mm × 5 mm. A finer mesh was applied to the corner regions of the chord members for greater precision.

C. Loading plates

The model incorporates rigid upper and lower-end plates. The

finite element analysis was performed using a three-dimensional (3D) mesh with four-node bilinear rigid quadrilateral shell elements (R3D4), as shown in Figure 1. For each specimen, a reference point (RP) was selected perpendicular to the plane of each end plate and directed toward the center of the plate, using the effective length between the pin ends. This reference point (RP) handled the boundary conditions, while the RP at the upper-end plate was responsible for applying the load. Figure 1 illustrates how a multipoint constraint (MPC) was used to link all degrees of freedom from the reference points (RP) to the column end nodes, simulating a rigid end-bearing plate.

A "Tie" constraint was used to connect the loading plate to the cold-formed steel channels, preventing rotations and displacements of the elements during loading. In ABAQUS, this tie constraint automatically combines the three-dimensional shell meshes.

D. Characteristics of materials

Experimental testing is essential for determining the material properties of the steel. For slender sections, the maximum stress mustn't exceed the yield stress. In finite element analysis, the stress-strain relationship is often simplified and approximated as bilinear or idealized for ease of modeling. This approximation is applied to the cross-section of the material. However, ABAQUS offers the flexibility to use a multilinear stress-strain curve, enhancing its capability to model various material behaviors.

In this parametric study, the properties of carbon steel were considered. The steel was modeled using von Mises isotropic hardening with a minimum yield stress (F_y) of 355 MPa and an ultimate strength (F_u) of 510 MPa. The shear modulus was set at 81,000 MPa. The bilinear elastic-plastic stress-strain curve with linear strain hardening was applied for the simulation. The linear elastic part of the curve was characterized by a Young's modulus (E_0) of 210 GPa and a Poisson's ratio of 0.3. In the nonlinear analysis, plastic hardening was incorporated into the model using the von Mises criterion, expanding the isotropic yield surface.

E. Boundary conditions and loads

The following is a breakdown of the structural analysis setup for columns with hinged ends.

1. Hinged Ends on CFS Battened Columns where the ends are hinged, meaning they allow rotation but no translation. This is crucial for modeling how the column behaves under different loading conditions.

2. Displacements and Rotations at Reference Points (RPs): Boundary conditions are applied at specific reference points (RPs) on the column to simulate these hinged ends. The displacements (translations) and rotations at these points are constrained in certain directions. The upper RP restricts translations along the X and Y axes and rotation about the Z-axis, while the lower RP has restrictions on all translations (X,

Y, Z) and rotation about the Z-axis.

3. Modified RIKS Technique in ABAQUS: using a nonlinear analysis method (Modified RIKS) within ABAQUS to incrementally apply the load and track the column's response, especially when dealing with large deformations. This method is useful for capturing post-buckling behavior and complex load paths.

4. Column Subjected to Beginning Moments ($M_{uy} = P_u * e_x$): The column is subjected to moments generated by an eccentric

axial load. The term ($M_{uy} = P_u * e_x$) suggests the creation of a bending moment due to the eccentricity (e_x) (the distance along the X-axis from the centerline of the column to the line of action of the axial load (P_u)).

The structural column is modeled under axial load with an eccentricity that generates bending. The boundary conditions reflect how the column is constrained at both ends to simulate hinged supports, and the Modified RIKS technique is used to capture the behavior under these loading conditions.

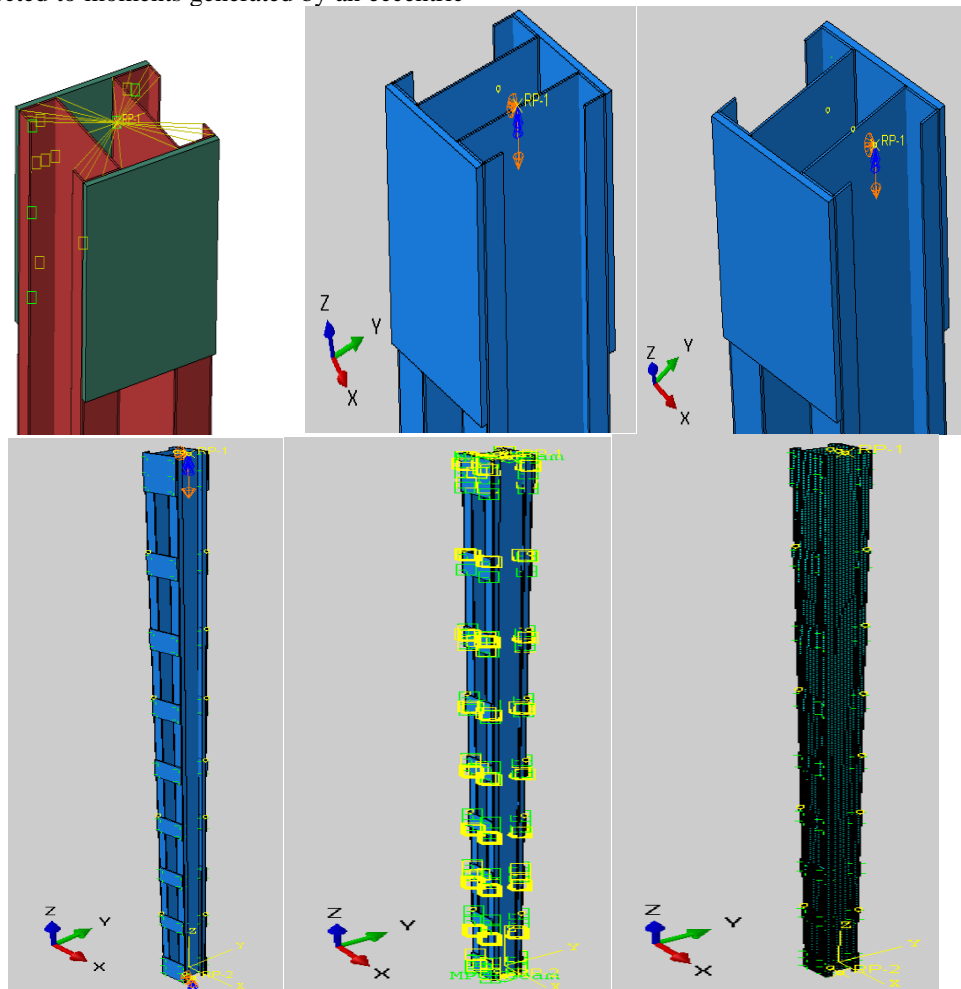


Figure1. Finite elements mesh, load application Fastener modeling, and Boundary conditions

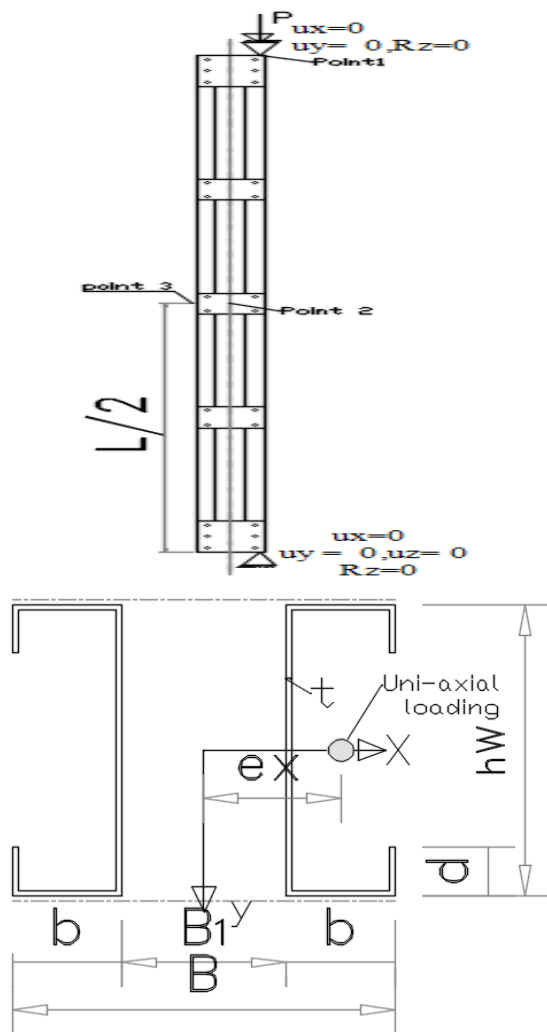


Figure2. Loading positions at column upper loaded end and support position at column lower support end for column subjected to Uni-axial loads

F. Initial geometric imperfections

The finite elements analysis considers the observed magnitudes of both failure modes: local buckling and overall buckling. Eigenvalue analyses of the column are performed to identify these buckling modes and establish the initial overall imperfections. According to the literature review [27-28], the average overall imperfections for the tested specimens were found to be 1/1000 of the total length of the specimen. These overall imperfections were simulated for battened columns made from cold-formed steel sections. All buckling modes predicted by the ABAQUS Eigenvalue analysis are normalized to 1.0, and the measured magnitudes of the initial local and overall geometric imperfections are incorporated into the buckling modes.

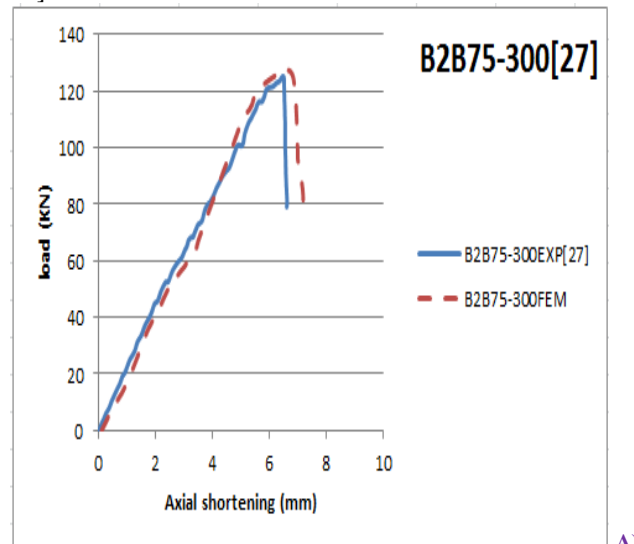
G. Verification of Finite Element Model

The nonlinear finite element model of the CFS open-section

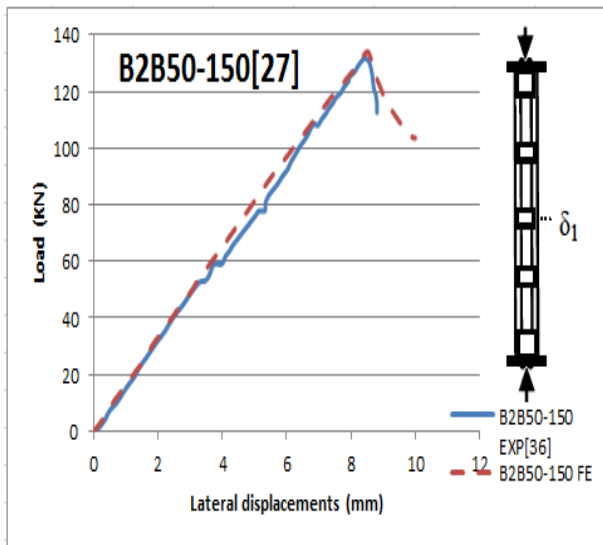
battened column, developed using ABAQUS [26], was calibrated against experimental data from previous studies [27, 28].

The experimental investigations of Dabaon et al. [27] focused on axially loaded built-up columns consisting two cold-formed steel channels placed back-to-back connected by batten plates with different longitudinal spacing. The column strengths, failure modes, load-lateral displacement, and load-axial shortening relationships predicted numerically were compared against those measured experimentally, as summarized in Table 1 and presented in Figs. 3. The column strengths obtained in the tests (P_{Test}) and finite element analysis (P_{FE}) were compared, as shown in Table 1 and presented in Figs. 3. El-Aghoury et al. [28] investigated a series of four equal slender angles subjected to uniaxial bending moments. The test specimens were made by connecting four CFS angle sections with batten plates. The column strengths obtained in the tests (P_{Test}) and finite element analysis (P_{FE}) were compared, as shown in Table 1 and presented in Figs. 4.

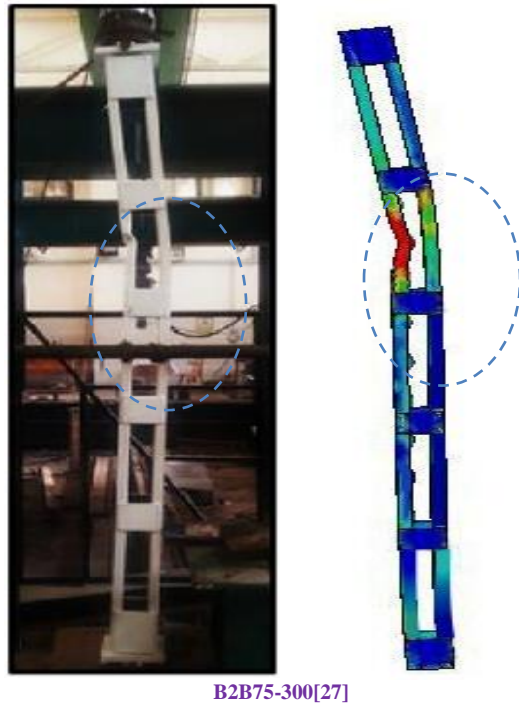
The CFS battened column finite element (FE) model was accomplished to be compared with the test results from these studies [27, 28]. Both failure modes and load-versus-lateral displacement curves were confirmed. Figures 3 and 4 display the load-lateral displacement, axial shortening, and failure modes. Table 1 compares the FEA and experimental ultimate capacities for Cold-Formed Steel (CFS) battened columns [27, 28].



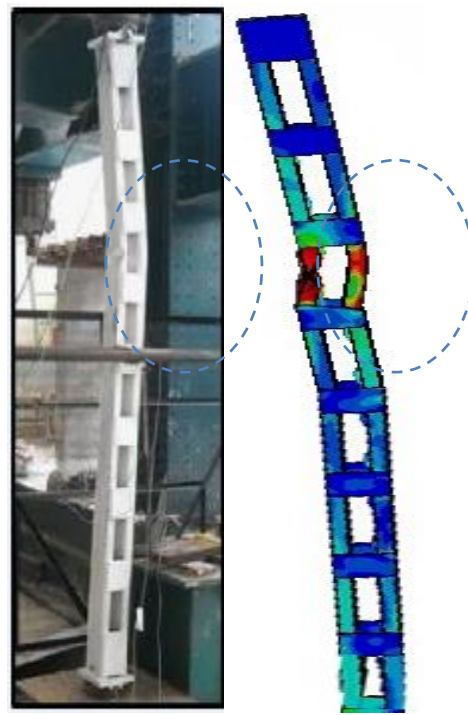
Axial shortening-load relationships



B) Lateral displacement-load relationships



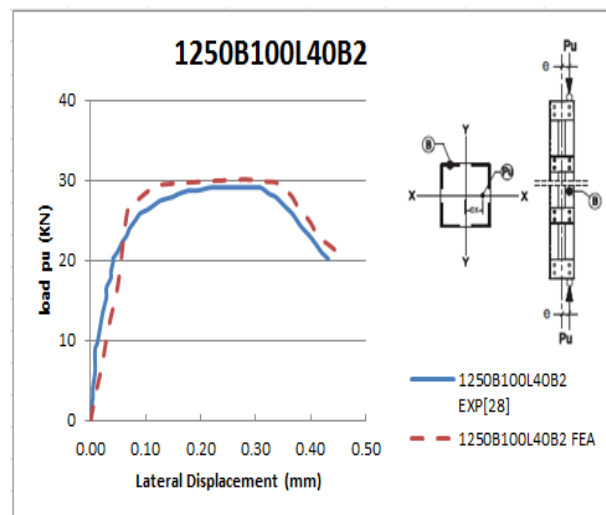
B2B75-300[27]

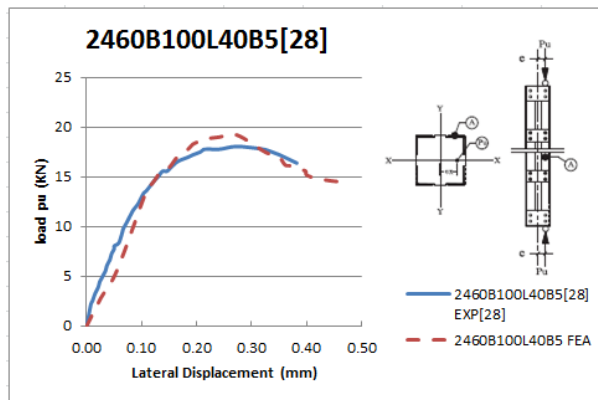


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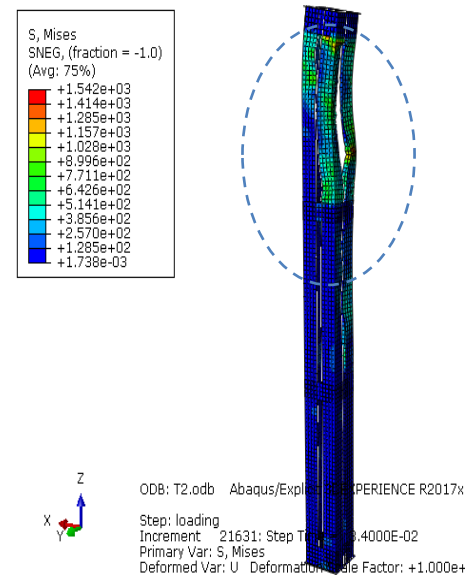
C) Comparison of the modes of failure between test specimens and the finite models of [27]

Fig. 3. (A): Lateral displacement-load relationships, (B): Axial shortening, and (C): failure modes Flexural buckling (FB) and local buckling (LB)





A) Lateral displacement-load relationships



B) Modes of failure shape of (1250B100L40B2) [28]

Fig. 4. (A): Lateral displacement-load relationships, and (B): failure modes
flexural-torsional Flexural buckling (coupled), flexural/flexural-torsional (FT)

Table1. Battered column Strength of the FEA and experimental test [27, 28]

Specimen	Load type	Load Ecc.		Exp. Results		FEA Results		P _{test} /P _{F.E.A}
		e _x mm	e _y mm	P _{test} (kN)	Failure mode	P _{F.E.A} (kN)	Failure mode	
B2B75–300 [27]	Axially load	-	-	125.2	LB	126.5	LB	0.989
B2B50–150 [27]	Axially load	-	-	133.12	FB	134	FB	0.993
1250B100L40B2 [28]	Uni-axially	30	-	28.2	FT	30	FT	0.933
2460B100L40B5[28]	Uni-axially	30		18	coupled	19.6	Coupled	0.918

120×40×20×2mm.

The comparison of FEA results with test results across the four aspects—strengths, failure modes, axial shortening, and the load vs. displacement relationship—shows strong agreement. Therefore, the FE model is considered suitable for the intended parametric study.

H. Parametric study overview

The parametric study analyzed 48 built-up cold-formed steel battered columns with varying apparent distances between the two channels (B1) and buckling lengths (L and L_z), as illustrated in Figure 5. Column dimensions are depicted in Figure 5 and detailed in Table 2. The columns had lengths of 2400, 3000, and 5800 mm and were categorized into four series, each with a back-to-back unobstructed distance (B1) of 40, 60, 80, and 90 mm. These series were subjected to uniaxial loading. The channel cross-section in (mm), h_w × b × d × t;

The buckling lengths (L_z) of single channels in each group were 600, 460, 450, 340, 250, and 175 mm. The corresponding slenderness ratios of the channels between batten plates (λ_z) were 38.7, 29.67, 29.03, 21.93, 16.12, and 11.29, respectively. The overall slenderness ratio ($\lambda = L/r_{min}$) of the columns, calculated according to AISI S-100:2016 [21], ranged from 50 to 162. The load eccentricities (U) ranged from 20 to 68 mm.

Specimen Labeling

The specimens were labeled as LB1-L_zU_{ex}, with an example label being "2400B40-450U20."

L: Column length – 2400 mm

B40: Indicates a 40 mm distance between the two channels in the built-up section (B1 = 40 mm).

- **U20:** Indicates uniaxial loading with a 20 mm eccentricity in the x-direction from the center of gravity.
- **450:** Denotes the buckling length (L_z) of a single channel along its minor axis, measured as 450 mm.

The ultimate strengths (P_{UFE}), modes of failure, and non-dimensional critical slenderness (λ_c) of the columns, as determined using North American specifications [21], are summarized in Table 3. The results were derived from the analyses of the finite element model developed for the built-up cold-formed steel battered columns.

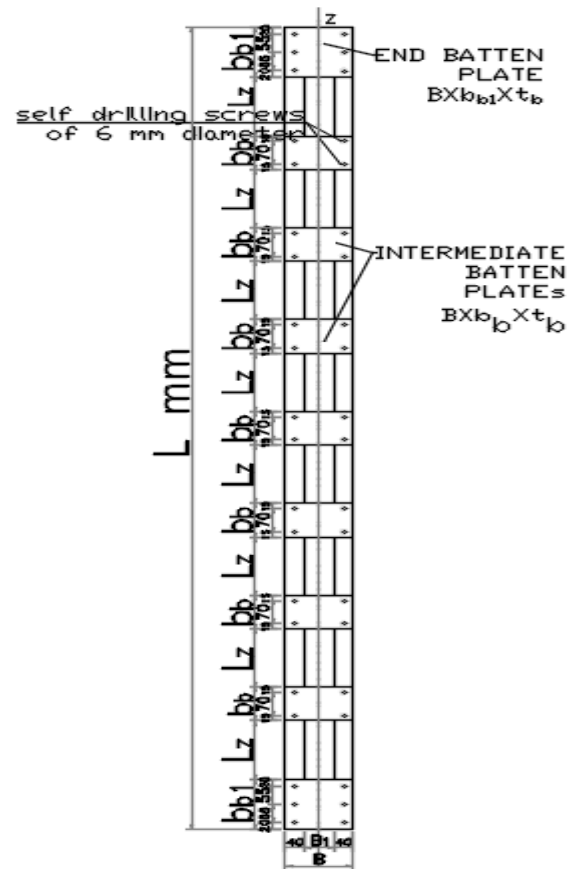
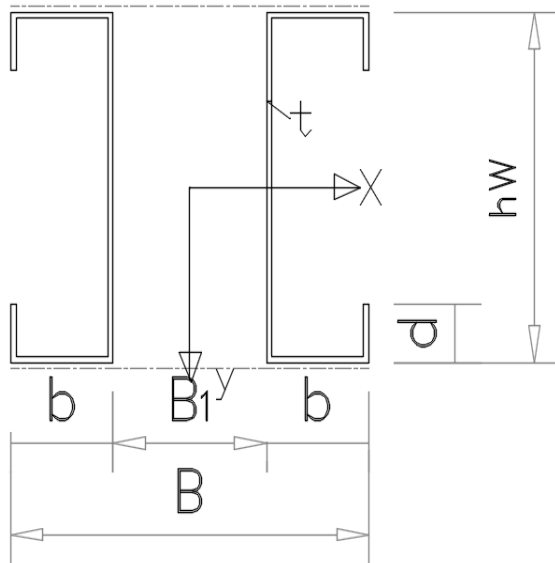


Fig.5.The cross-sectional dimensions analyzed in the parametric study.

Table 2. The parametric study's specimen dimensions.

Series	Specimen	L (mm)	Lz (mm)	B1/hw	λ_z	$\lambda = L/r_{min}$	Load eccentricity e_x (mm)	batten plates' dimensions in (mm)		
								b_b	b_{b1}	t_b
B40	2400B40-450U20	2400	450	0.33	29.032	71.29	20	100	150	6
	2400B40-450U48	2400	450	0.33	29.032	71.29	48	100	150	6
	2400B40-340U20	2400	340	0.33	21.935	68.71	20	100	150	6
	2400B40-340U48	2400	340	0.33	21.935	68.71	48	100	150	6
	2400B40-175U20	2400	175	0.33	11.29	66.08	20	100	150	6
	2400B40-175U48	2400	175	0.33	11.29	66.08	48	100	150	6
	3000B40-600U20	3000	600	0.33	38.70	90.20	20	100	150	6
	3000B40-600U48	3000	600	0.33	38.70	90.20	48	100	150	6
	3000B40-460U20	3000	460	0.33	29.67	86.60	20	100	150	6
	3000B40-460U48	3000	460	0.33	29.67	86.60	48	100	150	6
	3000B40-250U20	3000	250	0.33	16.12	82.90	20	100	150	6

	3000B40-250U48	3000	250	0.33	16.12	82.90	48	100	150	6
	5800B40-600U20	5800	600	0.33	38.70	162.1	20	100	150	6
	5800B40-600U48	5800	600	0.33	38.70	162.1	48	100	150	6
	5800B40-460U20	5800	460	0.33	29.67	160.1	20	100	150	6
	5800B40-460U48	5800	460	0.33	29.67	160.1	48	100	150	6
	5800B40-250U20	5800	250	0.33	16.12	158.1	20	100	150	6
	5800B40-250U48	5800	250	0.33	16.12	158.1	48	100	150	6
B60	2400B60-450U30	2400	450	0.5	29.032	59.60	30	100	150	6
	2400B60-450U56	2400	450	0.5	29.032	59.60	56	100	150	6
	2400B60-340U30	2400	340	0.5	21.935	56.49	30	100	150	6
	2400B60-340U56	2400	340	0.5	21.935	56.49	56	100	150	6
	2400B60-175U30	2400	175	0.5	11.29	53.27	30	100	150	6
	2400B60-175U56	2400	175	0.5	11.29	53.27	56	100	150	6
	3000B60-600U30	3000	600	0.5	38.70	75.71	30	100	150	6
	3000B60-600U56	3000	600	0.5	38.70	75.71	56	100	150	6
	3000B60-460U30	3000	460	0.5	29.67	71.52	30	100	150	6
	3000B60-460U56	3000	460	0.5	29.67	71.52	56	100	150	6
	3000B60-250U30	3000	250	0.5	16.12	67.04	30	100	150	6
	3000B60-250U56	3000	250	0.5	16.12	67.04	56	100	150	6
	5800B60-600U30	5800	600	0.5	38.70	131.6	30	100	150	6
	5800B60-600U56	5800	600	0.5	38.70	131.6	56	100	150	6
	5800B60-460U30	5800	460	0.5	29.67	129.2	30	100	150	6
	Specimen	L (mm)	Lz (mm)	B1/hw	λz	λ= L/r _{min}	Load Eccentricity e _x (mm)	batten plates' dimensions in (mm)		
								b _b	b _{b1}	t _b
	5800B60-460U56	5800	460	0.5	29.67	129.26	56	100	150	6
	5800B60-250U30	5800	250	0.5	16.12	126.84	30	100	150	6
	5800B60-250U56	5800	250	0.5	16.12	126.84	56	100	150	6
B80	2400B80-450U40	2400	450	0.67	29.03	51.99	40	100	150	6
	2400B80-450U64	2400	450	0.67	29.03	51.99	64	100	150	6
	2400B80-340U40	2400	340	0.67	21.93	51.94	40	100	150	6
	2400B80-340U64	2400	340	0.67	21.93	51.94	64	100	150	6
	2400B80-175U40	2400	175	0.67	11.29	51.94	40	100	150	6
	2400B80-175U64	2400	175	0.67	11.29	51.94	64	100	150	6
B90	2400B90-450U45	2400	450	0.75	29.03	51.94	20	100	150	6
	2400B90-450U68	2400	450	0.75	29.03	51.94	48	100	150	6

	2400B90-340U45	2400	340	0.75	21.93	51.94	20	100	150	6
	2400B90-340U68	2400	340	0.75	21.93	51.94	48	100	150	6
	2400B90-175U45	2400	175	0.75	11.29	51.94	20	100	150	6
	2400B90-175U68	2400	175	0.75	11.29	51.94	48	100	150	6

e_x : The eccentric distance along X-axis is equivalent to (0.5B1&0.4B) from the C. G. of the section

III. Design Method

A. Design rules following North American code (AISI S-100:2016) [21]

The design code NAS [21] presents two primary design methodologies: The Direct Strength Method (DSM) and the Effective Width Method (EWM). Accurately estimating the elastic critical buckling stress is a key factor influencing the precision and reliability of the DSM.

Built-up sections

For built-up compression members, the axial strength shall be determined following section E2 subjected to the following modifications.

$$\left(\frac{KL}{r}\right)_m = \sqrt{\left(\frac{kl}{r}\right)_o^2 + \left(\frac{L_z}{r_i}\right)^2} \quad (1)$$

Where:

$\left(\frac{kl}{r}\right)_o$ = overall slenderness ratio of the entire section about the built-up member axis

L_z = Intermediate fastener or spot weld spacing

r_i = minimum radius of gyration of the full unreduced cross-sectional area of individual shapes of built-up sections.

Uni-axially loaded column

- Calculation method in AISI S100-16

$$\frac{\bar{P}}{P_n} + \frac{M_y}{M_{ny}} \leq 1.0 \quad (2)$$

$$\bar{M}_y = B_1 e \bar{P} \quad (3)$$

$$B_1 = C_m / (1 - \alpha \bar{P} / P_{el}) \geq 1.0 \quad (4)$$

$$P_{el} = \pi^2 E I_y / (K_y L_y)^2 \quad (5)$$

Nominal flexural strength for lateral-torsional buckling as per the DSM

$$\text{for } M_{cre} < 0.56 M_y \quad (6)$$

$$M_{ne} = M_{cre}$$

$$\text{for } 2.78 M_y \geq M_{cre} \geq 0.56 M_y$$

$$M_{ne} = 10 / 9 M_y (1 - 10 M_y / 36 M_{cre}) \quad (7)$$

$$\text{for } M_{cre} > 2.78 M_y$$

$$M_{ne} = M_y \quad (8)$$

$$M_{nl} \quad M_{ne} \quad \lambda_1 \leq 0.776 \quad (9)$$

for

$$M_{ne} [1 - 0.15 (M_{cr1} / M_{ne})^{0.4}] (M_{cr1} / M_{ne})^{0.4}$$

$$\text{For } \lambda_1 > 0.776$$

(10)

Nominal flexural strength of distortional buckling by the DSM

$$\begin{cases} M_{nd} S_F f_y & \lambda_d \leq 0.673 \\ M_y [1 - 0.22 (M_{crd} / M_y)^{0.5}] (M_{crd} / M_y)^{0.5} & \text{for } \lambda_d > 0.673 \end{cases} \quad (11)$$

$$M_{crd} = S_F F_{crd} \quad (12)$$

$$\lambda_1 = \sqrt{M_y / M_{cr1}}, \lambda_d = \sqrt{M_y / M_{crd}} \quad (13)$$

$$M_n \text{ min } (M_{ne}, M_{nl}, M_{nd})$$

Where \bar{P} is the required compressive axial Strength; P_n is the available axial strength; F_{crd} is the elastic distortional buckling stress calculated by Appendix2; M_{cre} is the critical elastic local buckling moment nominal flexural strength of local yielding by the DSM in accordance with Appendix2; \bar{M} and M are the required flexural strength about the x- and y-axis, respectively, $\alpha = 1$, $C_m = 0$, e is the eccentricity, S_f is the gross section modulus referenced to the extreme fibre at first yield, Moreover, M_{ny} is the available flexural strength about the y-axis.

B. Design regulations following the EN 1993-1-3:2006 [20]

The un-factored design strength for class 4 is determined by EC3 (BS EN1993-1-3) as follows:

$$P_{EC3} = \chi A_e f_y \quad (14)$$

$$\chi = 1 / (\phi + \sqrt{\phi^2 - \lambda^2}) \text{ for } \chi \leq 1.0 \quad (15)$$

$$\phi = 0.5 [1 + \alpha (\bar{\lambda} - 0.2) + \bar{\lambda}^2] \quad (16)$$

$$\bar{\lambda} = \sqrt{(A_{eff} f_y / N_{cr})} = (L_{cr} / i) \sqrt{(A_e f_y / \lambda_1 A_g)} \quad (17)$$

for Class 4 cross-sections

$$\lambda_1 = \pi \sqrt{(E / f_y)} \quad (18)$$

Where:

χ : the applicable buckling mode reduction factor.

N_{cr} : refers to the critical force for elastic design associated with specific buckling mode

A_e : the effective cross-sectional area.

F_y : A proof stress of 0.2% ($\sigma_{0.2}$) is the same as the yield stress.

" α " imperfection factor.

For a laterally unrestrained beam, the buckling resistance for

the design moment approximated calculated as $M_b, R_d = \chi_{LT} f_y W_y / \gamma_{M1}$ (19)

$$\chi_{LT} = 1 / (\phi_{LT} + \sqrt{(\phi_{LT}^2 - \bar{\lambda}_{LT}^2)}) \text{ for } \chi \leq 1.0$$

Where:

χ_{LT} =lateral –torsional buckling reduction factor

W_y represents the suitable section modulus:
cross sections for class 4

$W_y = W_{eff, y}$;

γ_{M1} =partial factor =1

$$(N_{Ed} / N_{b, Rd})^{0.8} + (M_{Ed} / M_{b, Rd})^{0.8} \leq 1.0 \quad (20)$$

Where:

M_b, R_d : is the design buckle resistance moment;

M_{Ed} : is the design moment value.

Beam-column design by EN1993-1-1 and EN1993-1-3:

N_b, R_d : is the design buckling resistance of the compression member;

N_{Ed} : is the design value of compression force by EN1993-1-1 and EN1993-1-3.

Alternately, the interaction formula may be applicable:

IV. Reliability analysis

A reliability study assessed the existing design criteria for cold-formed steel built-up battened columns. The reliability study was performed according to the ASCE Specifications

Commentary [31]. The reliability index (β) ranges from 2.5 to 3.5 for cold-formed steel structural components. The design criteria are regarded as reliable if the reliability index exceeds 2.5. The load combination of 1.2DL + 1.6LL, with resistance factors (ϕ) of $P = 0.85$ and $M_y = 0.95$, was used to calculate the reliability index (β) for NAS [21] and EC3 [20] specifications, where DL denotes dead load and LL denotes live load. The statistical parameters ($M_m = 1.10$, $F_m = 1.00$, $V_m = 0.10$, and $V_F = 0.05$) used in the calculation are obtained from the Commentary of the ASCE Specification for compression members.

The resistance factor of 0.85 was considered according to AISI S100 to calculate the Q_m/R_n . A trustworthy design process has a reliability index greater than 2.5. The higher the reliability index, the more reliable the prediction. AISI S100:2016 provides equations to determine the reliability index (β).

$$\beta = \frac{\ln(M_m F_m P_m \frac{R_n}{Q_m})}{\sqrt{V_M^2 + V_F^2 + V_P^2 + V_Q^2}}$$

where:

P_m =Represent the Mean ratio of the finite element determined moment to the predicted moment for the design codes and V_p are the corresponding COVs

M_m = Mean ratio of the actual yield stress to the minimum specified value.

F_m = Mean ratio of the actual section modulus to the specified (nominal) value.

V_Q is also the mean load effect variation coefficient, which can be computed from the load ratio (in this study, 0.21)

Table 3. Comparison of FEM results with the design codes' ultimate column strength

Series	Specimen name	FE Results		Design strength				FE/Design	
		P_{uFE} (KN)	Failure Mode	P_{NAS} (KN)	λ_c	P_{EC3} (KN)	$\bar{\lambda}$	P_{uFE} / P_{NAS}	P_{uFE} / P_{EC3}
B40	2400B40-450U20	151.08	F+D+LTB	126.4	0.93	119.4	0.85	1.193	1.265
	2400B40-450U48	100.45	F+D+LTB	83.4	0.93	77.1	0.85	1.204	1.302
	2400B40-340U20	153.05	F+ D+ L	128.1	0.89	119.4	0.85	1.194	1.281
	2400B40-340U48	102.54	F+ L	84.1	0.89	77.1	0.85	1.218	1.330
	2400B40-175U20	157.48	F+ D+ L	129.6	0.86	119.4	0.85	1.215	1.321
	2400B40-175U48	108.3	F+LTB	84.8	0.86	77.1	0.85	1.277	1.405
	3000B40-600U20	112	F+L+LTB+D	106.0	1.17	100.6	1.06	1.056	1.113
	3000B40-600U48	75.9	F+L+D+LTB	72.9	1.17	68.35	1.06	1.040	1.110
	3000B40-460U20	119	F+L+D+LTB	108.2	1.13	100.6	1.06	1.099	1.182
	3000B40-460U48	77.2	F+L+D+LTB	74.0	1.13	68.35	1.06	1.043	1.129

	3000B40-250U20	122.3	F+ L+ D	110.3	1.08	100.6	1.06	1.108	1.215
	3000B40-250U48	79.8	F+ L+ D	75.0	1.08	68.35	1.06	1.063	1.167
	5800B40-600U20	41.927	L+F+LTB	40.1	2.12	38.75	2.05	1.043	1.081
	5800B40-600U48	31.53	F+LTB	31.0	2.12	30.83	2.05	1.015	1.022
	5800B40-460U20	43.08	L+F+LTB	41.3	2.09	38.75	2.05	1.042	1.111
	5800B40-460U48	33.74	F+LTB	32.7	2.09	30.83	2.05	1.032	1.094
	5800B40-250U20	44.93	L+F+LTB	42.4	2.07	38.7	2.05	1.058	1.159
	5800B40-250U48	35.76	F+L+LTB	34.4	2.07	30.8	2.05	1.039	1.159
B60	2400B60-450U30	155.7	F+LTB	134.2	0.78	122.6	0.68	1.161	1.27
	2400B60-450U56	110.02	L+F+D	95.1	0.78	88.3	0.68	1.172	1.301
	2400B60-340U30	157.6	F+D	135.9	0.73	122.6	0.68	1.160	1.285
	2400B60-340U56	116.82	D+L+F	99.1	0.73	88.3	0.68	1.172	1.314
	2400B601750U30	163.4	F+LTB	137.4	0.69	122.6	0.68	1.189	1.332
	2400B60-175U56	122.9	F+LTB+D	100.0	0.69	88.3	0.68	1.229	1.391
	3000B60-600U30	119.23	F+L+LTB+D	117.3	0.99	105.7	0.85	1.016	1.128
	3000B60-600U56	91.64	F+L+D+LTB	88.0	0.99	78.72	0.85	1.041	1.164
	3000B60-460U30	119.6	F+L+D+LTB	118.6	0.93	105.7	0.85	1.008	1.131
	3000B60-460U56	93.37	F+L+D+LTB	88.7	0.93	78.7	0.85	1.052	1.186
	3000B60-250U30	124.9	F+L+D+LTB	121.0	0.87	105.7	0.85	1.031	1.181
	3000B60-250U56	95.6	F+L+D	90.4	0.87	78.7	0.85	1.05	1.214
	5800B60-600U30	62.38	F+L	50.31	1.72	48.2	1.64	1.239	1.292
	5800B60-600U56	51.1	F+L	42.89	1.72	40.8	1.64	1.191	1.252
	5800B60-460U30	64.93	F+L+LTB	52.78	1.69	48.2	1.64	1.230	1.345
	5800B60-460U56	53.4	F+L+LTB	44.64	1.96	40.8	1.64	1.196	1.308
	5800B60-250U30	67.39	F+L	54.79	1.66	48.27	1.64	1.229	1.396
	5800B60-250U56	55.08	F+L	46.05	1.66	40.8	1.64	1.196	1.350
B80	2400B80-450U40	158.8	F+LTB	139.6	0.68	119.8	0.67	1.137	1.324
	2400B80-450U64	125.8	F+L	109.7	0.68	93.8	0.67	1.140	1.341
	2400B80-340U40	160.2	F+LTB	141.3	0.63	119.8	0.67	1.130	1.336
	2400B80-340U64	127.32	F+L	110.	0.63	93.8	0.67	1.148	1.357
	2400B801750U40	165.3	F+LTB	139.6	0.67	119.8	0.67	1.180	1.379
	2400B80-175U64	132.32	L+D	109.8	0.67	93.8	0.67	1.200	1.410
B90	2400B90-450U45	159.4	L+F	140.3	0.67	119.0	0.67	1.135	1.339
	2400B90-450U68	130.19	LTB+L	113.7	0.67	96.1	0.67	1.144	1.353

	2400B90-340U45	160.9	F+D	140.3	0.67	119.0	0.67	1.140	1.351
	2400B90-340U68	131.45	F+L	113.7	0.67	96.1	0.67	1.150	1.366
	2400B90-175U45	165.5	L+F	140.3	0.67	119.	0.67	1.170	1.390
	2400B90-175U68	135.8	L+LTB	113.7	0.67	96.1	0.67	1.190	1.412
Mean Pm = Average of the array(PFEM/Pcode)								1.133	1.263
COV, VP								0.074	0.105
Reliability index, β								3.32	3.545
Resistance factor, Φ								P=0.85&My=0.95	

L: local buckling

D: distortional buckling

LTB: lateral torsional buckling

F: flexural buckling

Table 3 compares the finite results with the design equations results for columns. The mean value and reliability index, according to DSM, are 1.133 and 3.32, but according to the EC3 interaction Equations, they are 1.263 and 3.545, respectively. Therefore, the current EC3(PEc3) predicts results that do not respect AISI specifications regarding the reliability index, while the Euro code buckling interaction DSM (PNas) predictions meet the NAS reliability requirement condition.

V. Discussion of the results

A. Interaction curves

Columns subjected to uniaxial loading, with varying channel slenderness ratios (λ_z) of 11.29, 16.12, 21.93, 29.03, 29.93, and 38.71 between batten plates, were analyzed for different column lengths and eccentricities. The interaction curves between normalized axial load ratios (P_u/P_y) and moment ratios (M_{uy}/M_y) were derived based on the results of the finite element analysis, as shown in Fig.6.

Column strength decreases with combined axial forces and moments, especially for higher slenderness ratios.

Shorter columns show higher normalized axial and moment ratios (P_u/P_y and M_{uy}/M_y) for the same slenderness ratios and back-to-back distances.

Reducing the slenderness ratio (λ_z) has minimal impact on strength for columns of the same length and back-to-back distance.

Increasing column length reduces load-bearing capacity.

The strength reduction due to combined axial and bending loads is more significant than for uniaxially loaded columns.

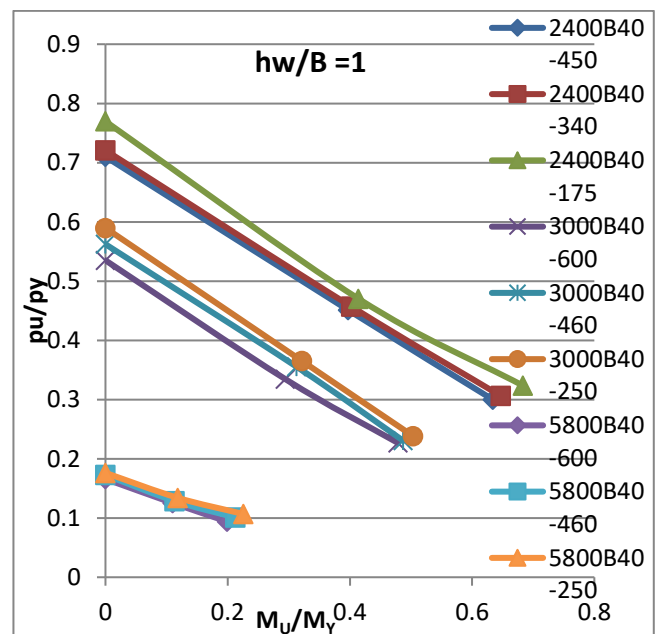


Figure 6: Interaction relationship curves for uni-axially loaded column

Comparison of the results obtained from FEA with the design codes.

This section as shown in figure 7, 8, and 9 compares the finite element's ultimate strength with the design rules outlined in EC3 [20] and AISI [21]. This comparison is summarized in Table 3 for uniaxial loading. The maximum design compressive force was calculated by incorporating compressive load values and eccentricities into axial compressive force-bending moment interaction equations. To ensure the strength of beam-column structures, the reliability index (β) was maintained within a range of 2.5 to 3.5, ensuring safe and reliable performance under varying loading conditions. This approach provides engineers with confidence in the adequacy of the design rules for achieving

sufficient safety margins.

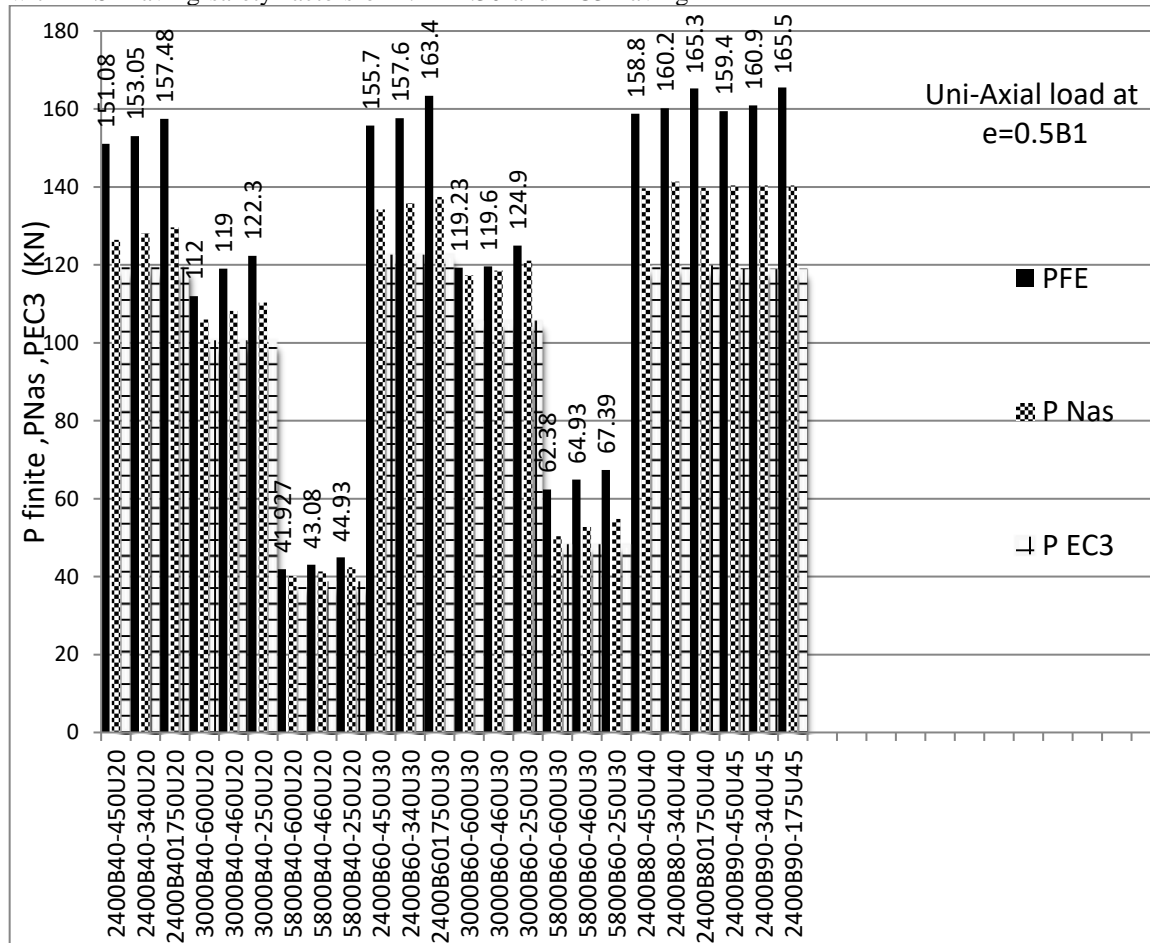
Slenderness ratio (λz) significantly affects the strength of medium and slender battered columns, with higher λz causing notable strength reductions.

Uniaxially loaded columns showed linear behavior, with peak axial strength decreasing as λz increased. For example, at a length of 2400 mm, strength decreased by 18% when λz raised from 11.29 to 21.9 and by 30% when λz increased to 29.03.

AISI and EC3 design codes provided conservative predictions, with AISI having safety factors of 1.14–1.30 and EC3 having

factors of 1.20–1.50. EC3 aligned more closely with finite element analysis (FEA), particularly for axial and uniaxial loads.

EC3 accounted for imperfections and effective widths, offering reliable predictions for medium-battered columns. AISI was more cautious for columns with uniaxial loads and eccentricity. Both codes generally yielded conservative results, with some cases showing strong agreement with FEA.



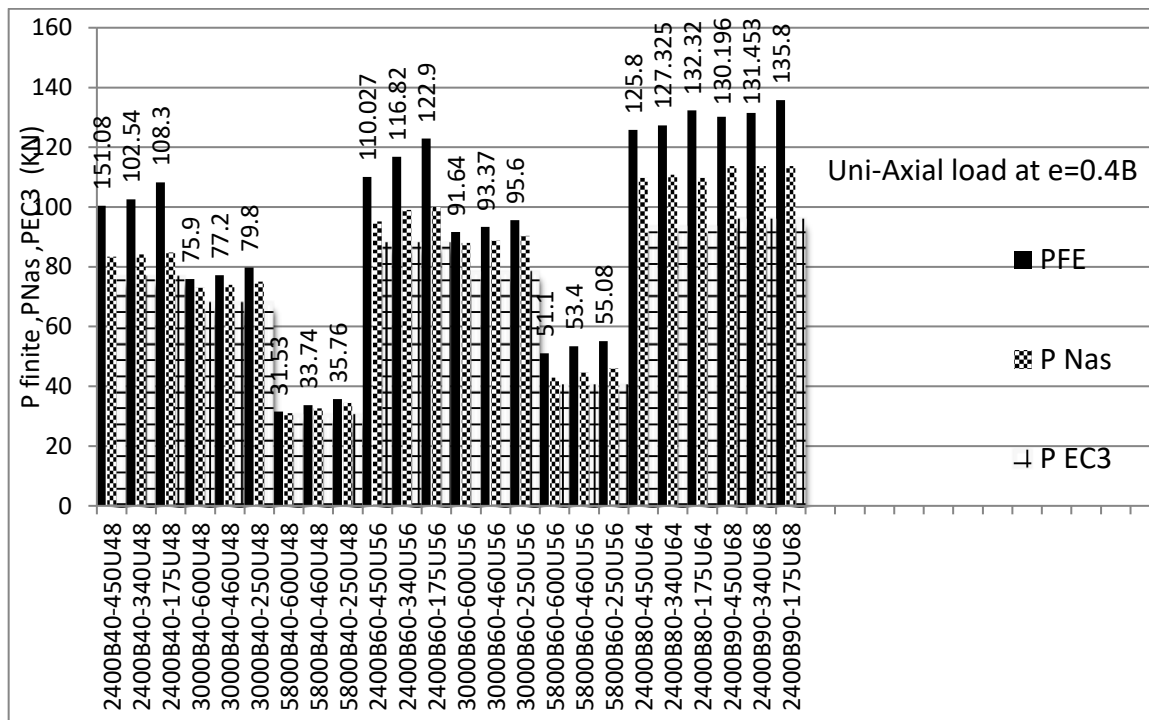
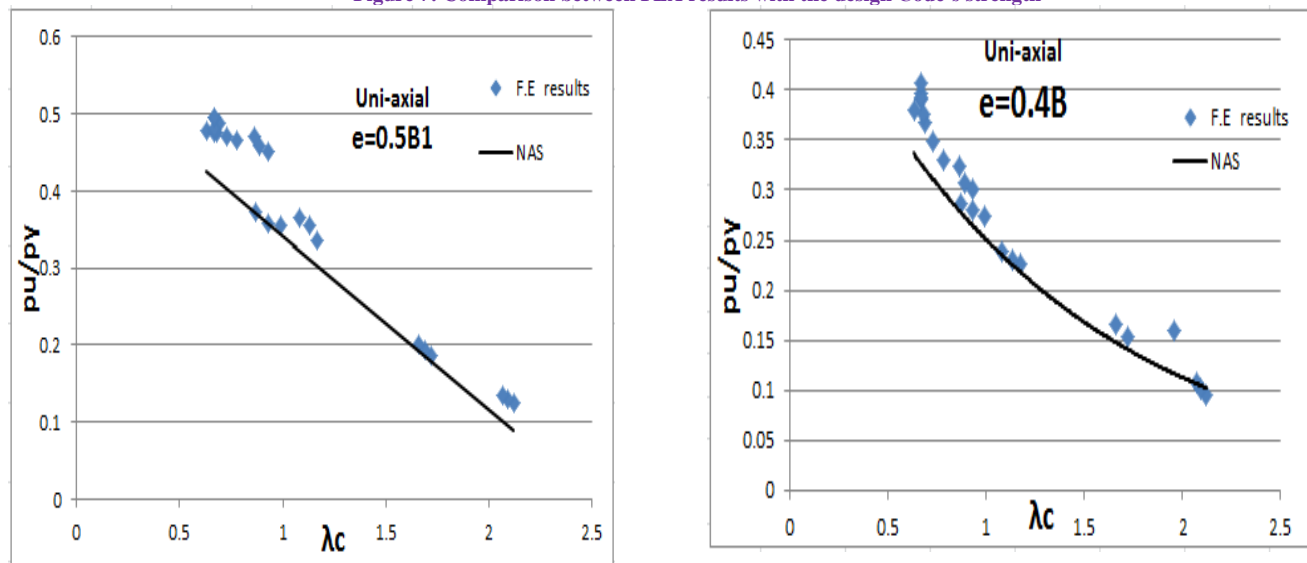


Figure 7: Comparison between FEA results with the design Code's strength

Fig. 8. The ratio of (p_u/p_y) the normalized axial forces to the non-dimensional slenderness (λ_c) NAS [21]

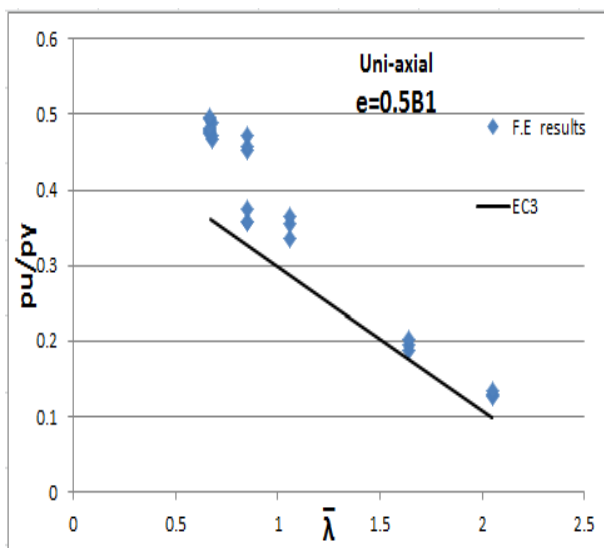
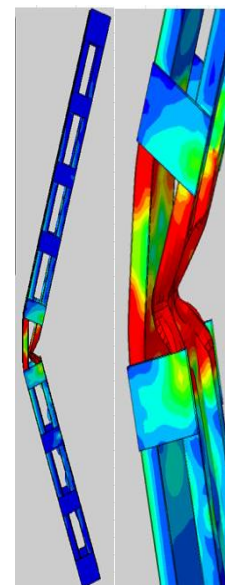
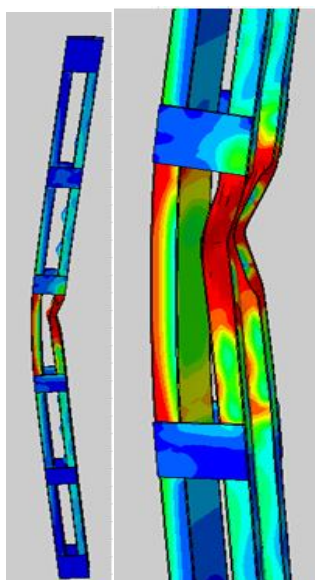
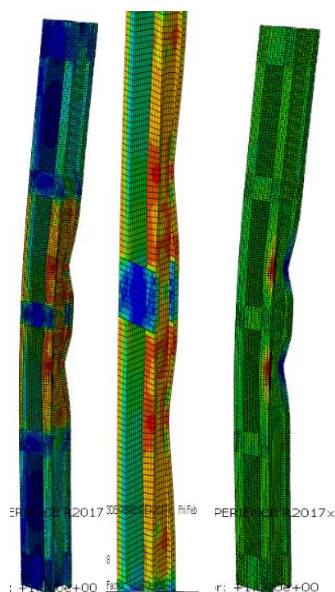
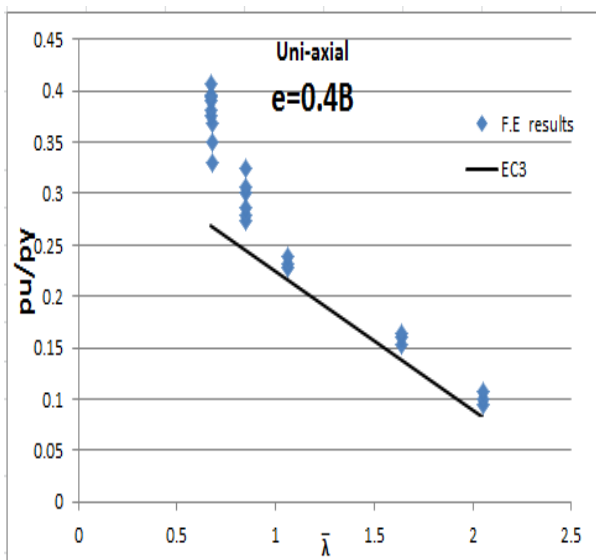


Fig. 9. The ratio of (p_u/p_y) the normalized axial forces to the non-dimensional slenderness for EC3[20] ($\bar{\lambda}$)

B. Failure modes' shape

For columns subjected to uni-axial loads, four primary failure modes occur due to buckling: local buckling, flexural buckling, distortional buckling, and lateral torsional buckling, which results from the buckling mode of shapes. Table 3 comprehensively summarizes the maximum load and the specific failure scenarios associated with each model.

In column model 2400B90-340U68, which had an ultimate load of 131.32KN, the waves of local buckling at the failure shape were seen to be more prominent at the column's web and flange and occur overall buckling of the section. However, for models 2400B40-340U20, 3000B60-250U30, 3000B40-250U48, B90A-340, and 2400B90-340U45, an interaction between flexural and distortional buckling was observed.



2400B90-

340U68 (F+L)

3000B60-460U56(F+L+D+LTB)

3000B60-250U30(F+L+D+LTB)

Fig.10.The failure shape

For column with slenderness ratios less than 1.1, failure is primarily dominated by local buckling, followed by distortional buckling modes. For higher slenderness ratios, failure is governed by a combination of distortional and flexural buckling modes.

The load-carrying capacity of built-up batten sections is largely influenced by the number of battens, as specified in design code provisions, and the spacing between the channels.

D. Axial shortening versus axial load

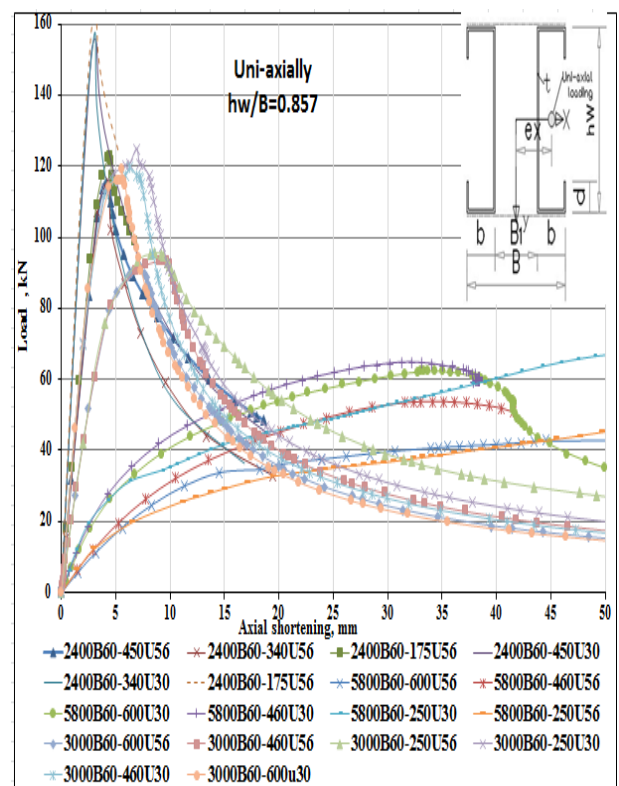
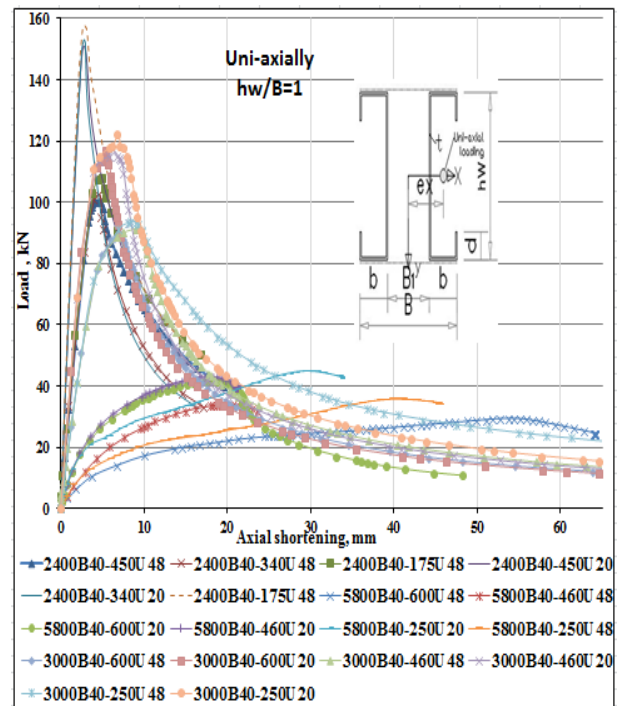
The magnitude of the axial compression was measured at point 1, located at the loaded end of the column, as depicted in the finite element in Fig.2. Figure 11 describes the relationship between the uniaxial load and axial shortening of all cold-formed steel (CFS).

Figures 11 show the relationship between axial shortening and axial load response for columns with the same depth-to-width ratio (hw/B). As the overall slenderness ratio increases, the maximum strength of the columns decreases. The plots for uniaxial load versus axial shortening display only a slight nonlinearity before the columns reach their peak axial strength. After reaching peak behavior, the columns experience a gradual decrease in uniaxial strength. The failure mode of the columns is characterized by a combination of local and global (flexural) buckling, including lateral torsional buckling.

When the column depth-to-width ratio (hw/B) is ≤ 0.75 and the overall slenderness ratios are high, a significant nonlinear response is observed in the relationship between axial load and axial shortening.

For uniaxially loaded columns, when the eccentricity of the load (ex) is equal to $0.4B$, the peak axial strength and axial shortening of the columns are influenced by the depth-to-width ratio (hw/B) for the same cross-section. For example, columns with a (hw/B) ratio of 0.7058 showed reductions in peak axial strengths of 14%, 9%, and 8% for slenderness ratios of $\lambda_z = 450$, 340, and 175, respectively, when the (hw/B) ratio increased from 0.7058 to 0.857. The reductions were 30%, 28%, and 25% when the (hw/B) ratio increased from 0.75 to 1.0 for $\lambda_z = 450$, 340, and 175, respectively.

Similarly, when the eccentricity of the load (ex) is equal to $0.5B$, columns with (hw/B) of 0.7058 showed reductions in peak axial strengths of 2.5%, 1.9%, and 1.2% for $\lambda_z = 450$, 340, and 175, respectively, when the (hw/B) ratio increased from 0.7058 to 0.857. The reductions were 4.7%, 4%, and 5% when the (hw/B) ratio increased from 0.75 to 1.0 for $\lambda_z = 450$, 340, and 175, respectively.



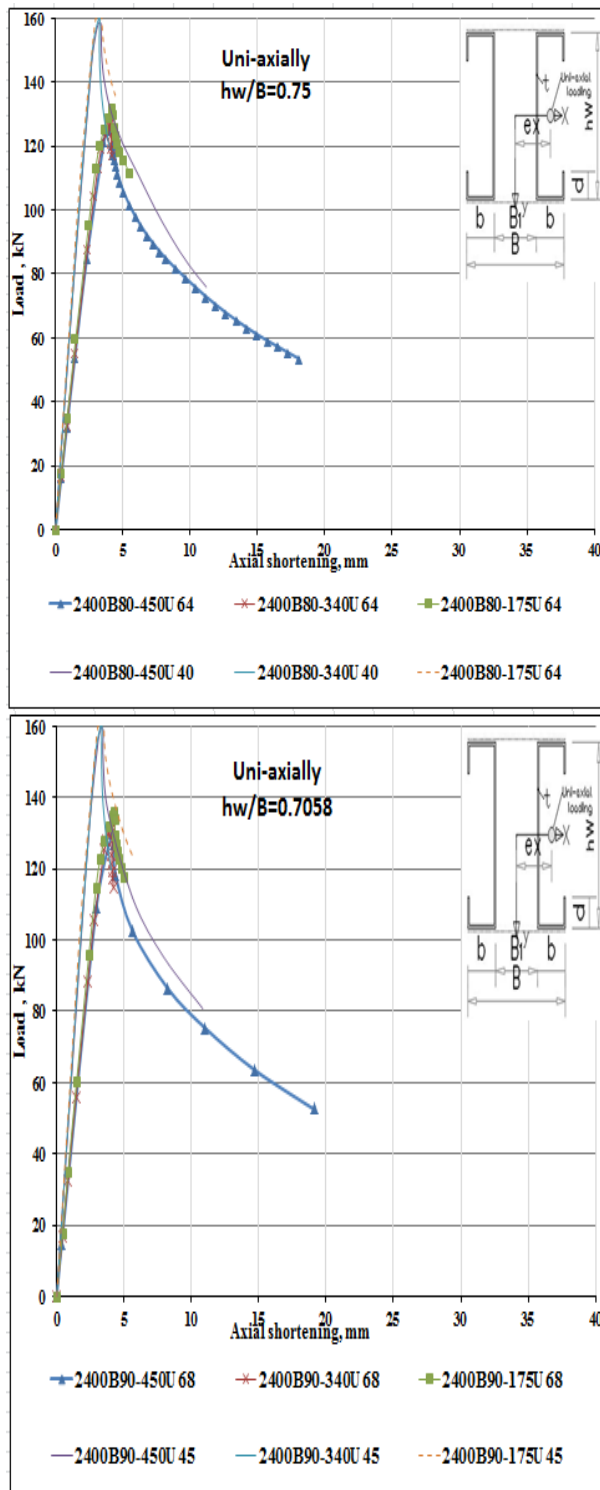


Figure 11: Relationships between load and axial shortening for uni-axially loaded columns

E. Load and lateral displacement (U_x and U_y)

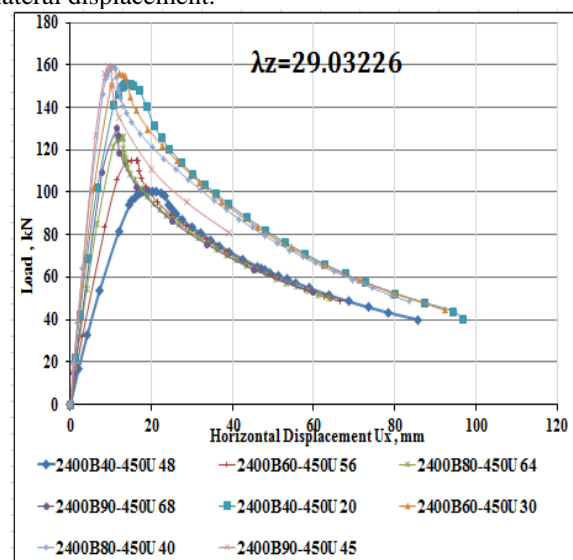
Lateral displacements were measured at two points along the column: "point 2" at the midpoint of the column and "point 3" at the midpoint of the cross-sectional flange, as shown in Figure 2.

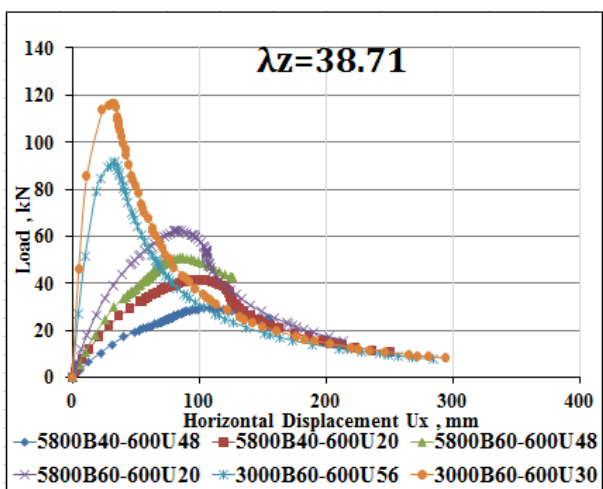
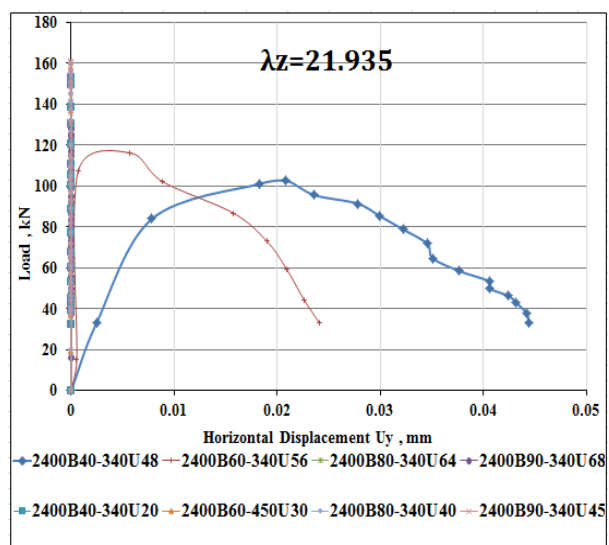
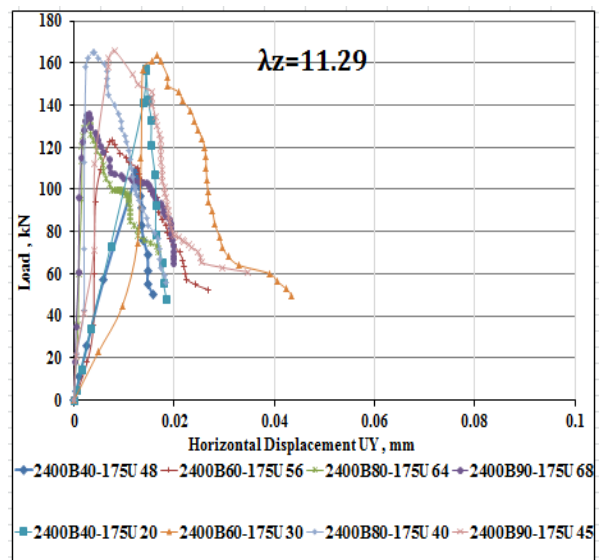
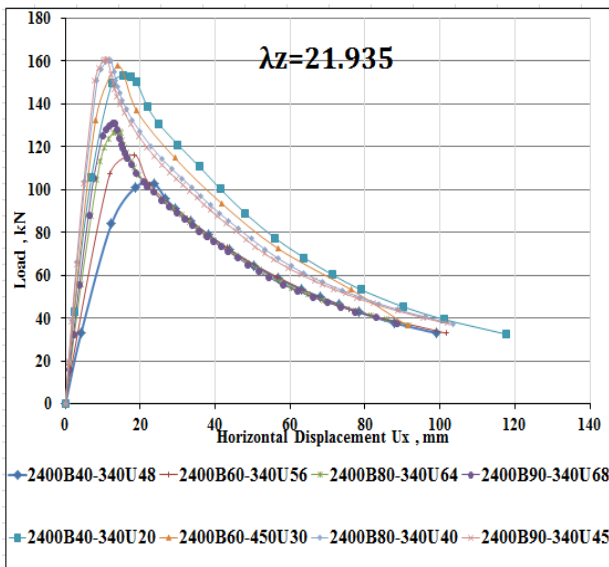
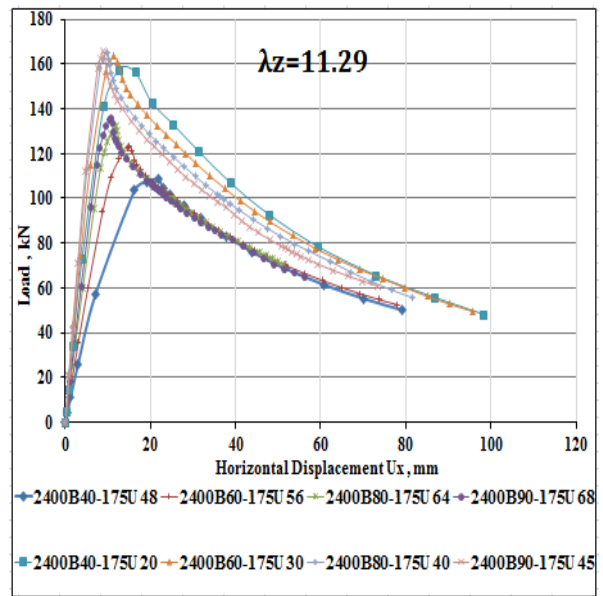
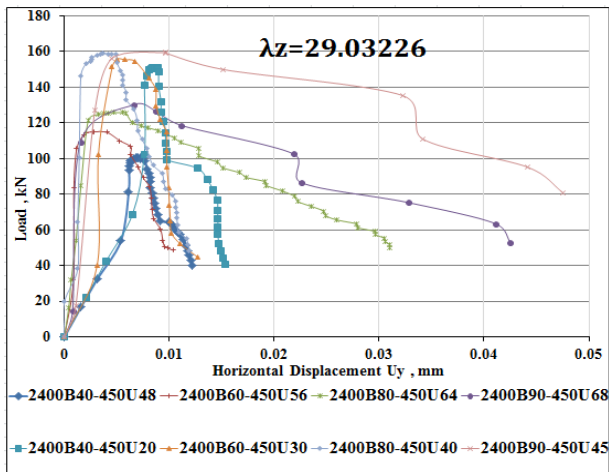
For a uni-axially loaded column, the lateral displacement in the x-direction (U_x) was more pronounced than in the y-direction (U_y). Figure 12 illustrates the relationship between the uni-axial load and lateral displacement response for columns with different channel slenderness ratios (λ_z). After short columns reach their peak strength, their strength drops significantly.

Columns with higher slenderness ratios exhibited better ductility compared to those with lower slenderness ratios. Thus, column ductility was directly related to the overall slenderness ratio. The lateral displacements (Δ) at which the columns reached their peak uni-axial strength were normalized by the column height (L).

The axial stiffness of a column under uni-axial loads was more evident for columns with a depth-to-width ratio (hw/B) of 0.7058 and 0.75. For columns with the same depth-to-width ratio, axial stiffness initially increased with the overall slenderness ratio. Additionally, columns with lower channel slenderness ratios ($\lambda_z = 11.29$) exhibited higher axial stiffness compared to those with slenderness ratios of 21.94 and 29.03.

Built-up columns made from unstiffened CFS channel sections were more prone to local buckling failures due to the absence of stiffeners in the chord members. As a result, the normalized lateral displacements of these columns showed a scattered upward trend. Furthermore, increasing the channel slenderness ratio (λ_z) resulted in higher lateral drift before failure and significantly reduced the scattering in lateral displacement.





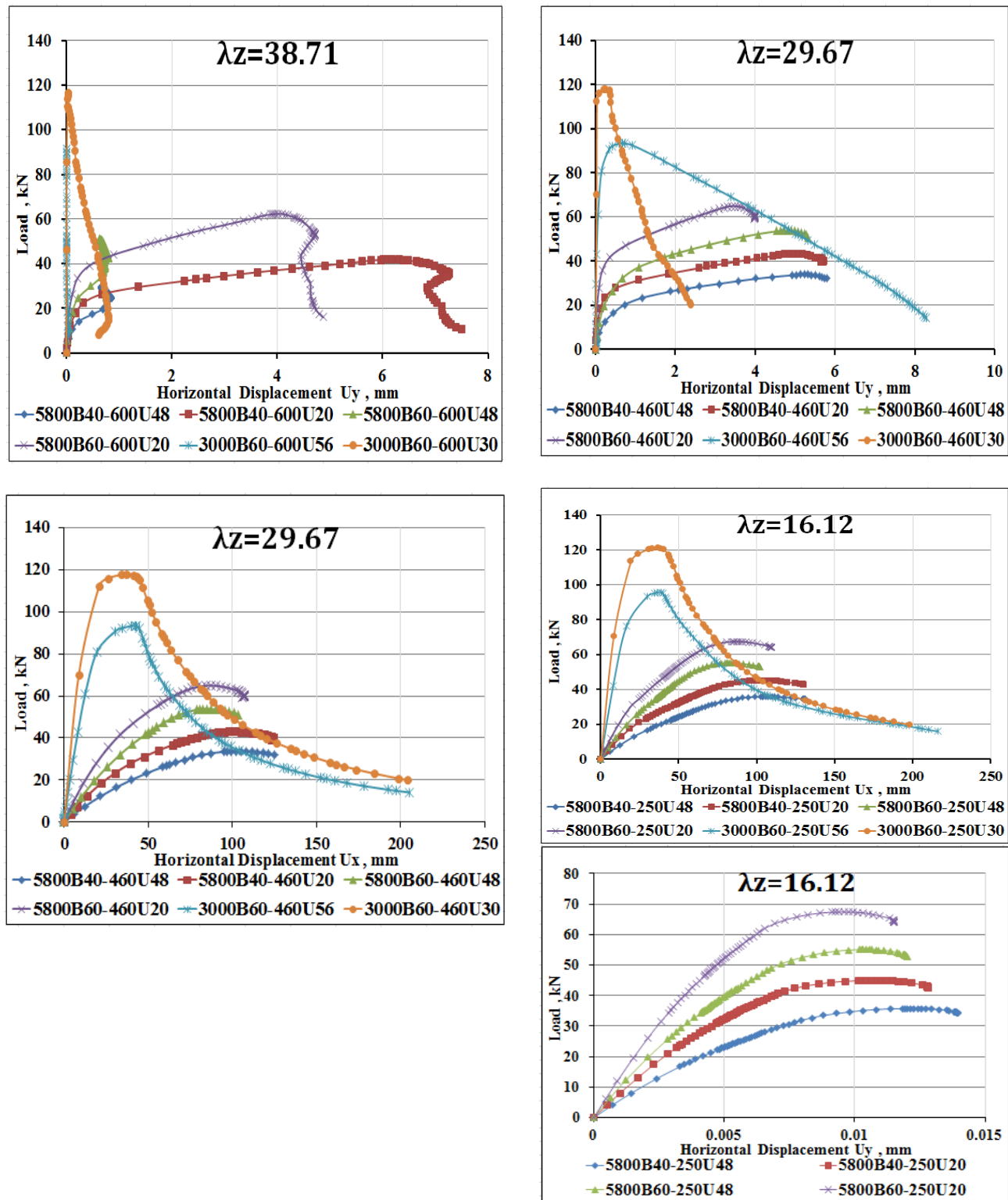


Figure 12: Relationships between load-induced lateral displacements (U_x) and (U_y) for columns subjected to uniaxial loading.

VI. Conclusions

The study used a nonlinear finite element model to evaluate

the ultimate capacity of battened beam columns, composed of two cold-formed battened channel sections in a back-to-back arrangement. The model accounted for both geometrical and material nonlinearities. For uniaxially loaded columns, local buckling is the primary failure mode for short columns, while long columns fail due to flexural buckling, accompanied by local buckling. Intermediate columns fail due to a combination of local and flexural buckling (interactive buckling).

The Eurocode3 and AISI LRFD predictions for uniaxially loaded battened columns were found to be conservative, with Eurocode3 being less reliable than AISI. Initial geometrical imperfections ($\Delta = L/1000$) were included in all loading cases, even for axially loaded columns. Axially loaded columns showed the highest cross-sectional capacity, in contrast to uniaxially loaded columns, which exhibited significant flange buckling in the local channel for both short and medium columns.

The DSM approach's results were conservative for columns that fail due to local buckling, while they were slightly conservative for columns that fail due to global buckling. According to EC3 specifications, open sections are considered conservative for columns failing in local buckling and slightly conservative for those failing in global buckling. The interaction curves between axial load and moment take a concave downward shape for columns with intermediate slenderness ratios. Additionally, the study found that the ultimate strength is reduced by increasing the slenderness ratio of the channel between batten plates (λ_z) or by decreasing the back-to-back distance between the channel sections.

Conflicts of Interest: The authors declare no conflict of interest.

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